

Improvement of Expected Performance of MONSTER onboard HiZ-GUNDAM satellite

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ABSTRACT

- Current development status of MONSTER, a visible & near-IR telescope onboard HiZ-GUNDAM
- Possible improvement of expected scientific performance of MONSTER, namely the detector system

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List of my talk

1. Introduction - purpose of MONSTER onboard the GRB search satellite
2. Current Design Specifications of MONSTER
3. Key technical challenges in MONSTER development
4. Status of on-going works
 - (1) BBM development of MONSTER Telescope
 - (2) Imaging sensors
5. Possible improvement of scientific performances

Summary

HiZ-GUNDAM satellite

Wide field X-ray Monitor

EAGLE Exploration of Ancient GRBs with Lobster Eye



Satellite Bus (Attitude control.
Data Handling, telecom. etc.)

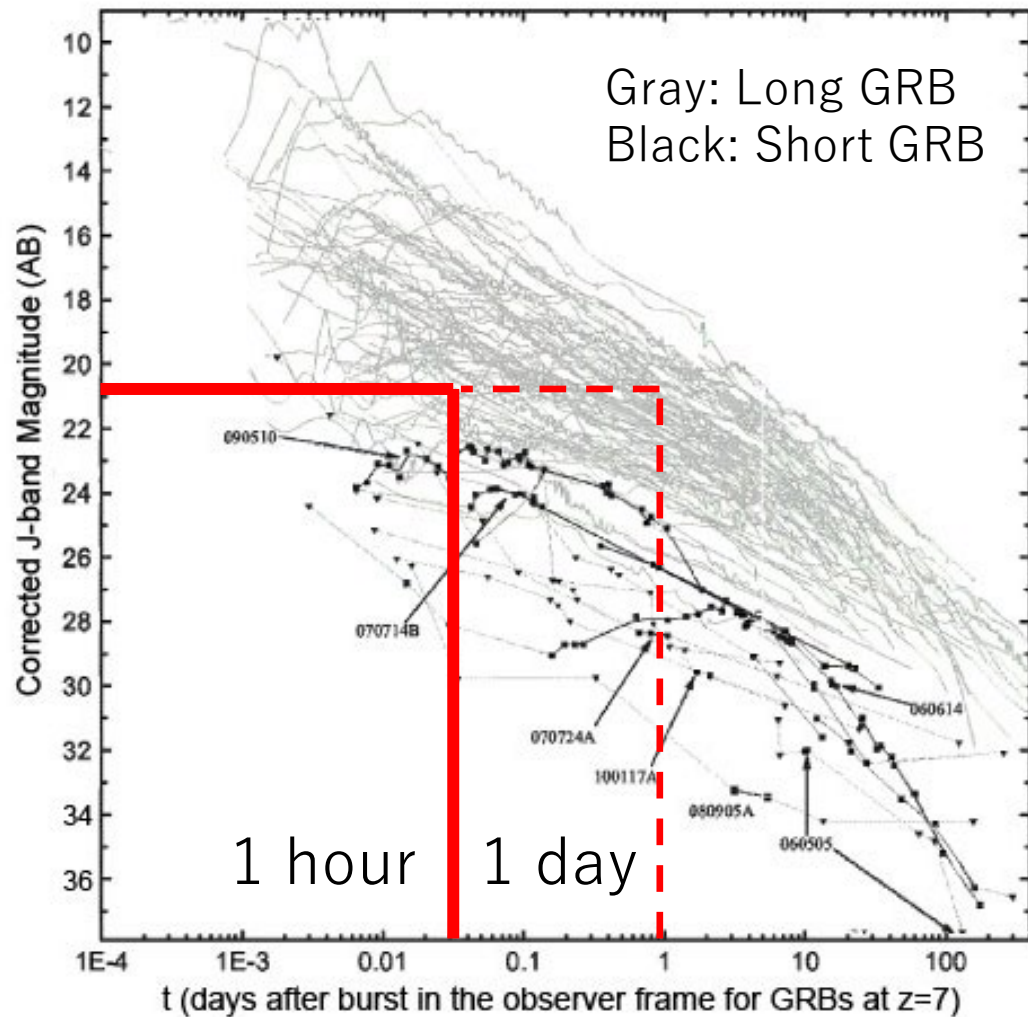
Visible and near-IR telescope

MONSTER Multiband Optical and Near-IR
Simultaneous Telescope for Efficient Response

1. Introduction

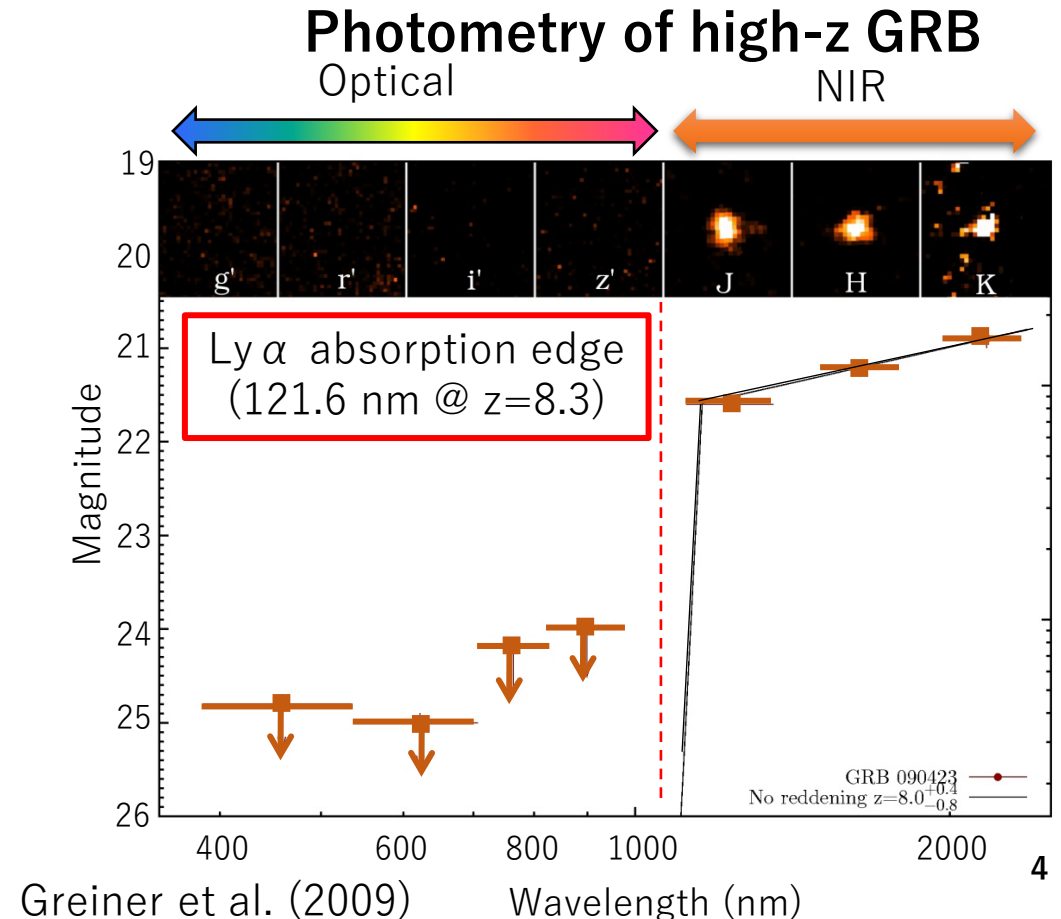
Quick follow-up is a Key to catch high- z GRB

Expected Light curve for GRBs @ $z = 7$



Based on Kann et al. (2010)

The afterglow is expected to be brighter than **21mag at 1 hour** after the burst (in $>90\%$ probability)



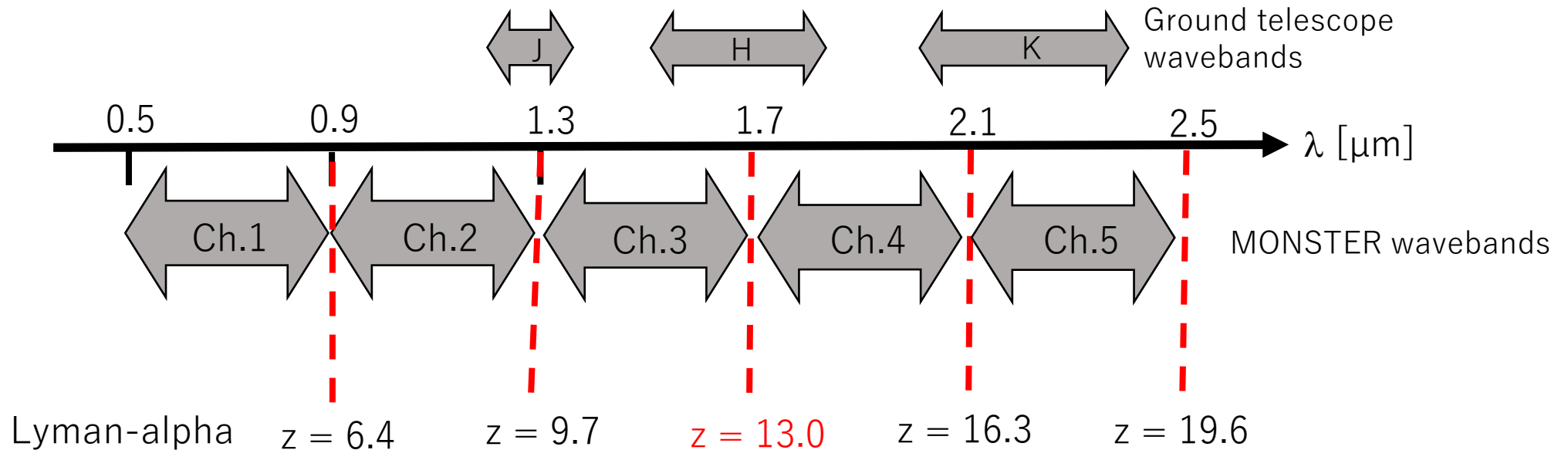
Greiner et al. (2009)

Wavelength (nm)

1. Introduction

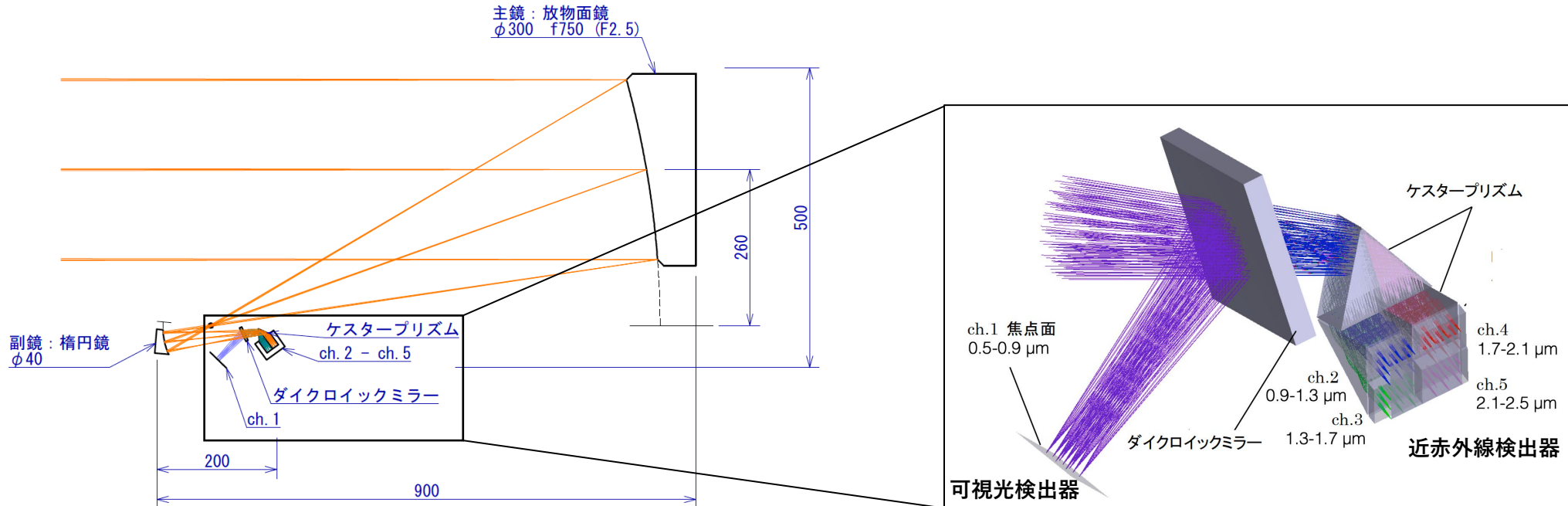
Purpose of MONSTER onboard the GRB search satellite

- **Rapid (goal: within 15 min) automatic follow-up** toward target detected by EAGLE
- **Identify GRB afterglow & estimate its (crude) photometric redshift**
 - Redshift can be determined with good accuracy up to $z < 13.0$.
 - In principle, detection is possible up to $z < 19.6$



2. Current Design Specifications of MONSTER

- A reflective (offset Gregorian) telescope with 30 cm aperture
- Simultaneous 5-band (visible + 4 NIR) imaging via dichroic mirror and a **Kösters prism**
- Field of View: 15 x 15 arcmin² (12 x 12 un-vignetted) : sufficiently larger than EAGLE error circle (3 arcmin radius) two image sensors(visible and NIR) with 3 arcsec image quality
- Cooled Telescope : shall be below 200 K, while the NIR sensor is below 120 K by only passive, radiative cooling system
- Telescope structure: all aluminum, athermal design (similarly contraction by cooling)



3. Key technical challenges in MONSTER development

- Thermal design, with only passive cooling to cool and maintain telescope below 200 K, detectors below 120 K
 - industrial and inhouse study (see R. Kageyama's poster #62 for in-house study)
 - Optical: good image quality shall be maintained at cold
 - Verification by BBM with "athermal design"
 - Kösters prism development (see Hori's talk next)
 - a noble beam-splitter splitting incident light into multiple wavelengths on a single image sensor
 - Image sensors (detectors) with electronics
 - Onboard image data processing (see H. Niinuma's poster #57)
- With the financial support of *Transformative Research Areas (A)*, we expect to perform all of the above activities. Hereafter I'll introduce on:
- **BBM telescope development and test**
 - **Image Sensor development (commercial CMOS sensor for visible band)**

4. Status of on-going works

(1) BBM development of MONSTER Telescope

- Purpose

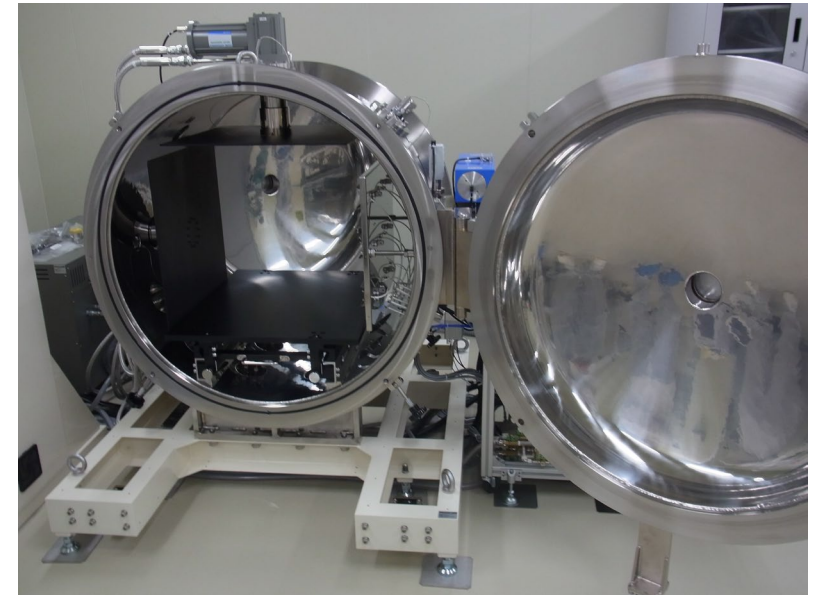
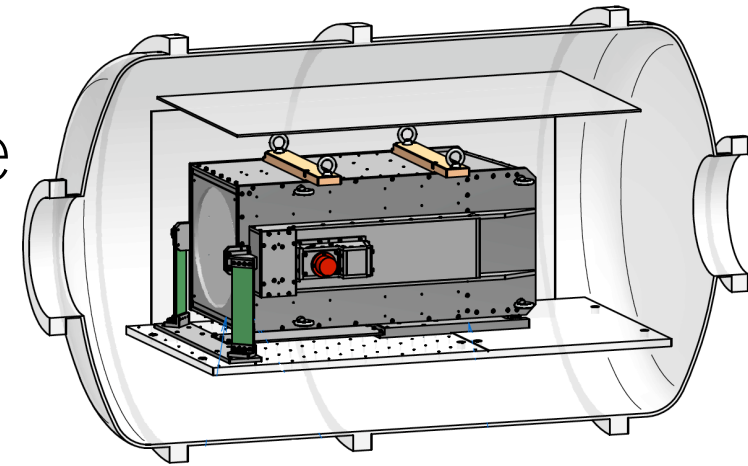
- Verify the cryogenic optical performance is within the requirement
 - demonstration of athermal design (all major structure components are made of Aluminum)

- Method

- Modify the existing cryo-vacuum chamber at Kanazawa University
- Image quality at cold is to be measured by introducing a $\sim 30\text{ cm } \phi$ collimated beam

- Planned Schedule

- FY24: mechanical and thermal design of test configuration
- FY25: design and fabrication of BBM mirrors and structures
- FY26: BBM test in the cryo-vacuum chamber at Kanazawa University



4. Status of on-going works

(2) Imaging sensors

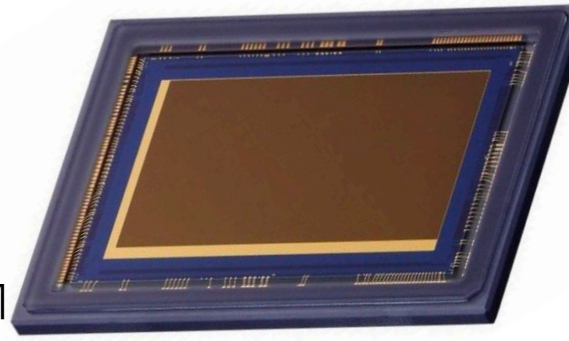
[Visible-band detector]

- Teledyne HyViSI or Canon LI3030SAM
- HyViSI is current baseline, but we also seek for the possibility to use commercially available C-MOS detectors
 - Canon LI3030SAM : Cryogenic and radiation tests completed, now onboard control/readout electronics is under study

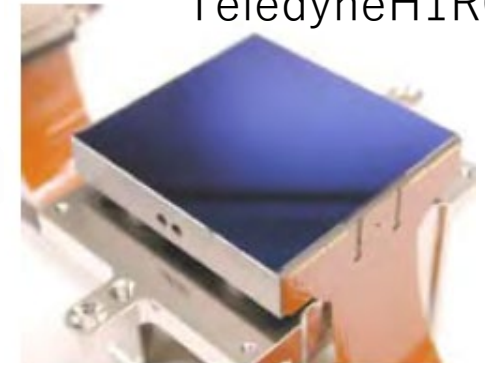
[Near-IR detector]

- Teledyne H1RG + SIDCAR (or ACADIA) ASIC
- Dark current must be $< 0.3 \text{ e}^-/\text{s}$ from sensitivity requirement
 - temperature $< 120 \text{ K}$

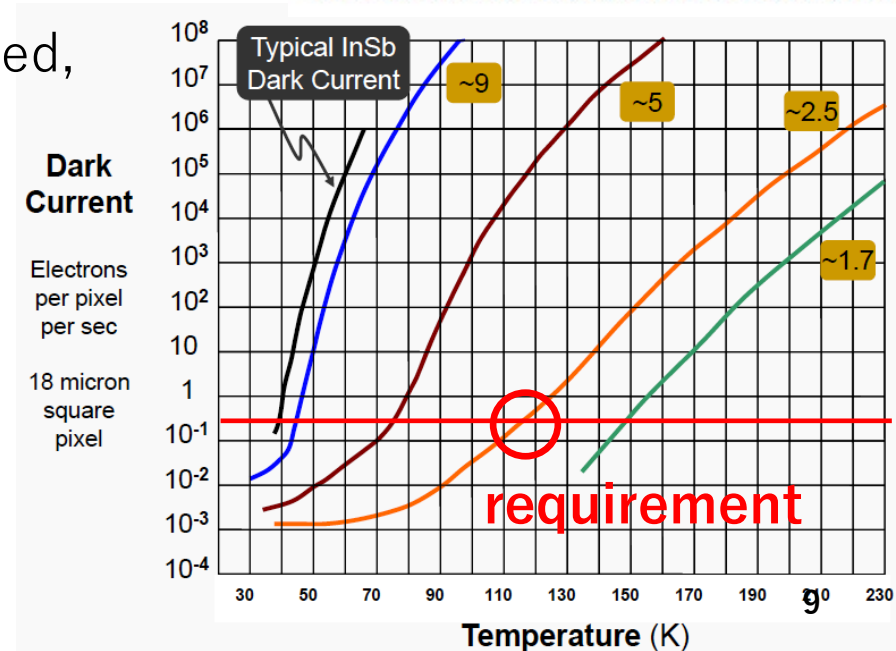
Canon LI3030SAM



TeledyneH1RG



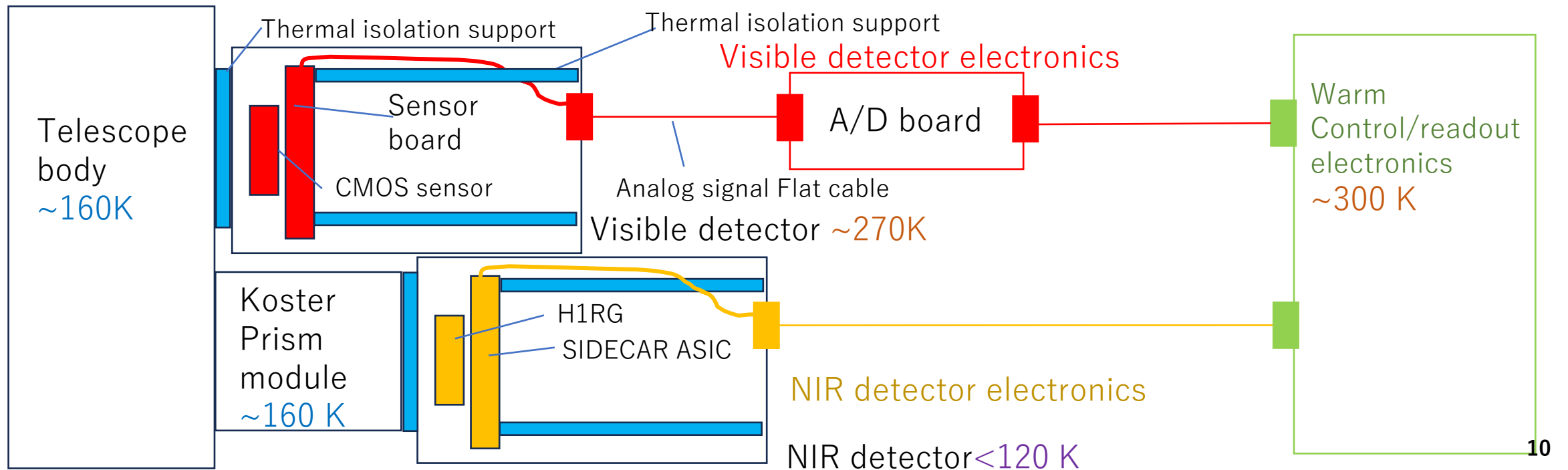
SIDCAR ASIC Focal Plane Electronics



4. Status of on-going works

(2) Imaging sensors : thermal/electrical architecture

- Challenges on commercial CMOS sensor onboard
 - Commercially used sensor normally provides us with fast (10~50 fps) readout but with large heat dissipation
 - but we only need relatively slow (0.1 – a few fps) readout but low heat dissipation → challenge in the readout/thermal design
 - In-house study (with help by two young JAXA members by a scheme of “intra-JAXA internship”)



5. Possible improvement of scientific performances

Through our current activities we hope to obtain:

- By BBM telescope development
 - After verifying athermal design, we will also get how the image quality changes with temperature gradient
 - how much can we accept the earthshine illumination to telescope
 - improvement of GRB follow-up capability
- By Visible band detector development
 - Challenges: slow readout/low heat dissipation, thermal isolation
 - Once successful, this is the first visible sensor readout system onboard JAXA astronomical satellite!
 - ⇒ Prospects to apply for future astronomy mission by JAXA
 - Obtain larger FoV for the visible channel:
 - LI3030SAM has 4-5 times larger format (with same pix size) than the HiViSI

HyViSI vs Commercially available visible sensors

	Teledyne HyViSI+SideCAR	Sony IMX533CLK	Gpixel GSENSE6060BSI	Cannon LI3030SAM
Pixel number (a)	1024 × 1024	3015 × 3080	6144 × 6144	2160 × 1286
Pixel size [μm] (b)	18	3.76	10.0	19
Size [mm] (a × b)	18.4 × 18.4		61.4 × 61.4	41.04 × 24.32
A/D conversion	16 bit	14 bit	14 bit	14 bit
Efficiency			95% @ 580 nm	86% @ 550 nm
Readout Noise [e-]	17		2.3 (HDR mode)	3.4 (rms, 40°C)
Dark current [e-/sec]	0.3		0.019 @ -55°C	300 @ 60°C
Output	Digital	Digital		Analog
Comment	Space Flight proven (e.g., JWST)	VERTEX (6U CubeSAT)	Operation guaranteed down to -55°C(218 K)	Space flight (nano-satellite) Tomo-e Gozen (@Kiso Obs.)
Baseline		Option		

Summary

- MONSTER onboard HiZ-GUNDAM will provide us with a rapid follow-up capability of targets detected by EAGLE
 - (1) LARGE FoV ($15 \times 15 \text{ arcmin}^2$)
 - (2) simultaneous 5-band imaging at $0.5 - 2.5 \mu\text{m}$
 - (3) Passively cooled (radiatively) telescope ($< 200 \text{ K}$), and a near-IR detector ($< 120 \text{ K}$)
- Possible improvement of science performances
 - BBM telescope \Rightarrow improvement of GRB follow-up capability
 - Visible detector \Rightarrow larger FoV for ch.1; application to future astronomy mission by JAXA
- Upcoming Plan (by FY2026)
 - Establish the thermal / mechanical design (in-house & industrial studies)
 - BBM optical performance test results will be available
 - Demonstration of CMOS visible sensor in space-use operation

