CAMELOT and IXPE: All-Sky Monitoring and Polarimetry for Multimessenger Astrophysics (3)

T. Mizuno (Hiroshima Univ.)
on behalf of the CAMELOT and IXPE team

2025 Nov. 19, Annual Conference on Multi Messenger Astrophysics @ Miyagi



Introduction: Role of X-ray Missions in MM Astrophys.



Z120r

3

役害

実績

アーカイブ

新着

女学会 2023年秋季年会@名古屋大学

(Adopted from M. Serino's slides at ASJ meeting, 2023.09)

X線のミッションが果たす役割は

広視野型 (Wide FOV)

- 重力波/ニュートリノと電磁波をつなぐ
 - 広視野の観測ができるので重力波/ニュートリノと同時観測ができる
- (重力波よりは)精度良い位置情報を 提供でき、その後の追跡観測につなが る
- 背景天体が比較的少なく、新天体をみつけやすいSimultaneous obs. with GW/neutrino events

- 追跡観測で放射源の詳細にせまる
 - イメージ、光度曲線、スペクトルが 同時にとれる
 - 高い時間分解能力、高いエネルギー 分解能、偏光など、特徴的な性能を持つ装置(衛星)がある

Imaging, photometry (light curve) spectroscopy, and polarimetry

(Follow-up) 詳細観測型

CAMELOT and

XPE



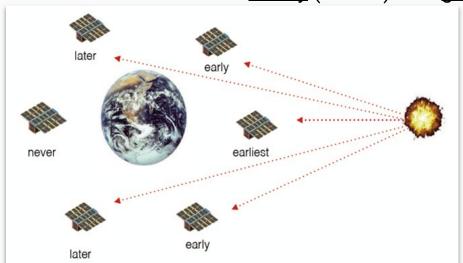
Wide FOV Obs. with a Fleet of Nano Sat.



Simultaneous observation using GW/neutrino and EM wave is essential for MM Astrophys (e.g., GW170817, IC-170922A)

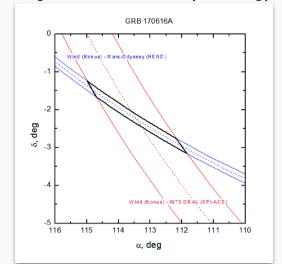
• We need to monitor all sky ($\sim 4\pi$ sr) with good angular resolution (<=1deg)

2025.11.19



~10 nano satellites allow all-sky monitoring in hard X-ray and gamma rays

T. Mizuno



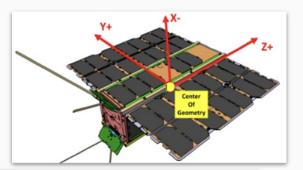
Timing-based localization; triangulation principle with different arrival time (<=0.1 ms) enables sub-degree localization 3/11

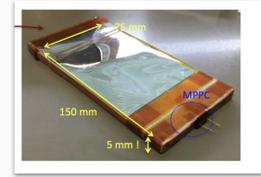


CAMELOT Mission

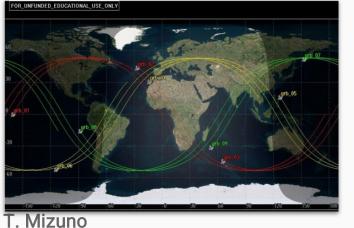


Hungarian-Japanese-Czech-Slovak project using standardized nanosatellites





Werner+18 Ohno+20



Satellite platform	3-U cubesat platform
Target orbit	>=9 satellites constellation in LEO with various orbital configuration
Payload	150x75x5 mm³ CsI readout by multi-channel MPPCs
Goal	Degree-scale timing-based localisation with a similar sensitivity to the Fermi-GBM

2025.11.19 4/11

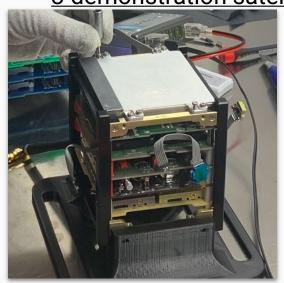


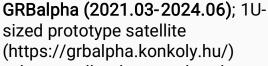
Mission Status



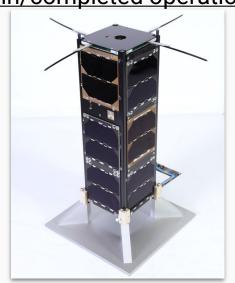
Pal+23; Ripa+22; Münz+24

3 demonstration satellites in/completed operation



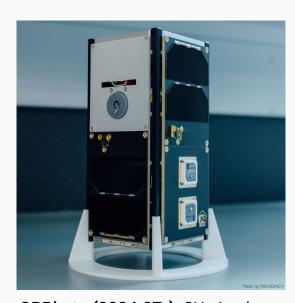


 has smaller detector, but the same basic concept
 T. Mizuno



VZLUSAT-2 (2022.01-); 2 prototype detectors as 2ndary payloads (https://www.vzlusat2.cz/en/)

 allows simultaneous GRB detection using nanosatellites 2025 11 19



GRBbeta (2024.07-); 2U-sized technological precursor (https://grbbeta.tuke.sk/index.php/en/)

• tests several technologies such as attitude control and UV camera

5/11



In-orbit Performance

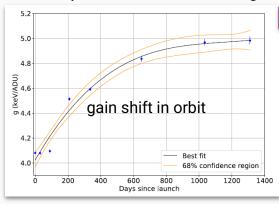


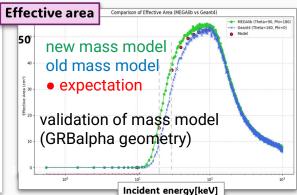
Pal+23; Ripa+22&25; Münz+24

GRBalpha completed >4 yrs operation successfully

- >200 transients detection
- Characterization of MPPC degradation in LEO (Ripa+25)

Mass model and detector response of GRBalpha&VZLUSAT-2 being developed (Yokota)





GRB beta: commissioning completed

- testing UV camera (precursor for future UV obs.) and attitude control
- also confirming transient detection (12(L)+1(S) GRBs since 2025.03)

Statistics of significant detection

(as of 2025.11.13)	GRBalpha	VZLUSAT-2
GRB (short)	127(23)	70(10)
Solar flare	101	74
SGR	2	3

Light curves of detected events are in public (one transient per \sim 5 day)



X-ray Polarimetry for Proving Mag. Field and Disk/Conona Geometry



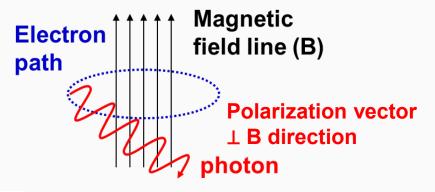
HE e- + B-field produce synchrotron radiation polarized $(\Pi_{\text{max}} = \frac{p+1}{p+7/3} \sim 0.7)$ on electron's orbital plane

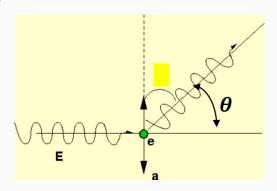
Unique probe for B-configuration (and accelerated electrons)

Scattered photos are polarized $(\pi = \frac{1 - (\cos \theta)^2}{1 + (\cos \theta)^2})$ along the plane of scattering

 Unique probe for geometry of disk/corona around compact objects

We can investigate B-field & disk/corona using X-ray polarimetry by IXPE





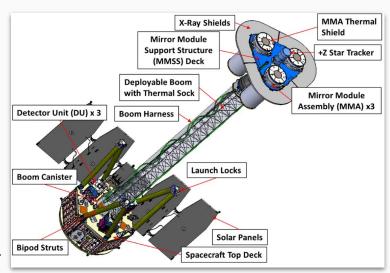


IXPE Mission



The first mission devoted to spatially-resolved X-ray polarimetry

- NASA SMEX mission, launched in 2021 Dec
 - Bilateral collaboration btw. NASA/MSFC & Italian Space Agency (w/ Japanese group providing key devices)
- 2 year mission (baseline) + Guest Observer Program (2024 Feb.-)
- Imaging-polarimetry in 2-8 keV for the first time
- Data are archived by NASA's HEASARC, released 1 week after the completion of the observation
- Unticipated ToO can be requested via IXPE website; latest example is that for Crab PSR after glitches (see also Goya's poster for PWN analysis)



3 x (Mirror + Gas Pixel Detector) FOV=12.9' x 12.9', HPD=25", μ_{100} >0.5 achieved (see Weisskopf 18)



Notes on Detector Performance in Orbit

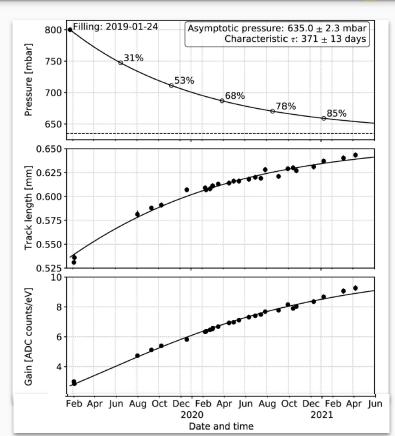


GPD pressure decrease gradually (known issue before the launch)

- will affect efficiency, track length, and gain (and instrumental response)
- We continuously monitor the efficiency and update responses

DU2 anomaly since 2025.04

- Part of readout ASIC not work
- We continuously take data, but need to recalibrate the spurious modulation correction
- <u>DU2 date (after 2025.05) removed temporary</u>.
 Stay tuned





Notes on Detector Performance in Orbit

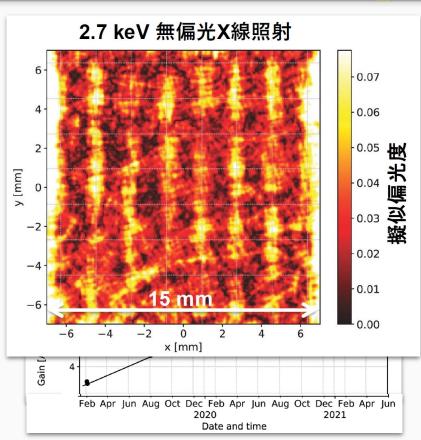


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Science w/ IXPE: Multi-λ Polarimetry of Mrk 421



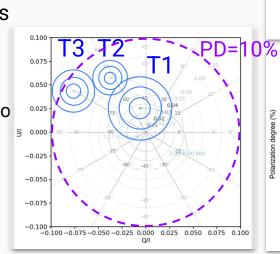
High-synchrotron-peaked blazar (at z=0.031)

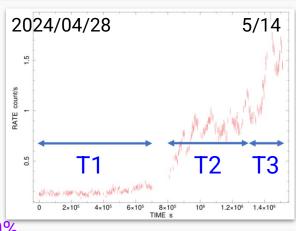
Flux increased by x10 in 2024; Coordinated polarimetry in X-ray (IXPE) and IR&opt. (Kanata)
(Tochihara+25 @ASJ meeting)

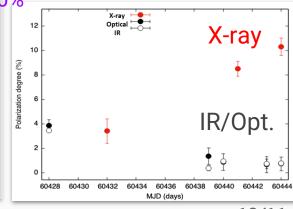
Significant change of PD_X by Q/U analysis

During the flare, only PD_X increased $(PD_{IR} \& PD_{opt} \text{ stayed low and stable})$

- HE particles are localized and unrelated to those producing IR/opt.
- PD_X of T1 is lowest among past observation; likely depends on the flux









Summary and Future Prospect



All-sky monitoring and X-ray polarimetry are crucial in multimessenger (and multiwavelength) astrophysics

CAMELOT mission enables all-sky monitoring in hard X-ray with good spatial resolution using fleet of nanosatellites

- Demonstration satellites; GRBalpha, VZLUSAT-2 and GRBbeta
 - GRBalpha completed operation successfully
 - One transient per ~5 day and light curves are publicly available (GRBalpha and VZLUSAT-2)
 - o GRBbeta was launched last year; testing technologies and detecting transients

IXPE is the first mission devoted to spatially-resolved polarimetry in soft X-rays

- Data are made public after completion of observation, ToO also possible upon request
- GPD pressure change and DU2 anomaly (no need to worry once you understand them)
- It reveals B-configuration of AGN, PWN, etc. (also reveals disk/corona geometry)



References & Useful Links



- GRBalpha transient light curves (https://monoceros.physics.muni.cz/hea/GRBAlpha/) and website (https://grbalpha.konkoly.hu/)
- VZLUSAT-2 transient light curves (https://monoceros.physics.muni.cz/hea/VZLUSAT-2/) and website (https://www.vzlusat2.cz/en/)
- GRBbeta website (https://grbbeta.tuke.sk/index.php/en/home/)
- Werner et al. 2018, Proc. SPIE 10669, 2;Ohno et al. 2020, Proc. SPIE 11454, 114541Z
- Pal et al. 2023, A&A 677, 40; Ripa et al. 2022, Proc. SPIE 12181, 121811K; Münz et al. 2024, Proc. SPIE 13093 130936J; Ripa et al. 2025, NIMA 1076, 170513
- IXPE Archive (https://heasarc.gsfc.nasa.gov/docs/ixpe/archive/)
- IXPE technical information (https://ixpe.msfc.nasa.gov/for scientists/index.html)
- Kislat et al. 2015, Astropart. Phys. 68, 45; Vink & Zhoug 2018, Galaxies 6, 46 (Stokes Param. analysis)
- Di Gesu et al. 2022, ApJL 938, 7; Di Gesu et al. 2023, Nature Astronomy 7, 1245; Kim et al. 2024,
 Maksym et al. 2025, ApJ 986, 230 (Mrk 421 results)
- Weisskopf 2018, Galaxies 6,33; Soffitta et al. 2021, AJ 162, 208; Baldini et al. 2021, Astropart.
 Phys. 133, 102628 (Mission & Instruments)

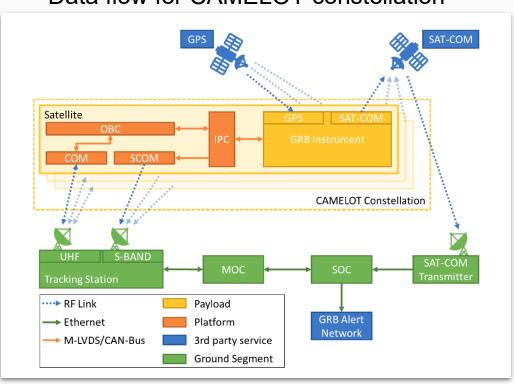
Backup Slide



Mission Concept



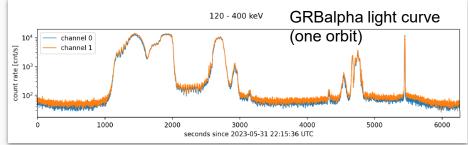
Data flow for CAMELOT constellation



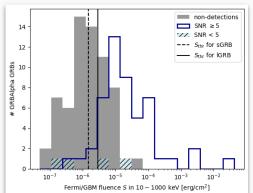
- Following on-board trigger, satellite payload will downlink data using global satellite communication module
- After the localization of GRB, the SOC will send GRB alert



In-orbit Performance



~2/3 of orbit is suitable for GRB detection



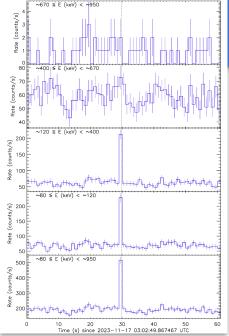
~70% Fermi-GBM GRBs (with fluence of >=3x10⁻⁶ erg/cm2) will be detected if in FOV and low background region

M. Dafčíková (Ph.D. thesis)

Also in progress: commissioning of GRB beta, characterizing BG In future: implement a rate trigger algorithm for autonomous detection T. Mizuno 2025.11.19

Pal+23; Ripa+22; Münz+24

Light curves of detected events are in public; one transient per ~5 day



Statistics of significant detection

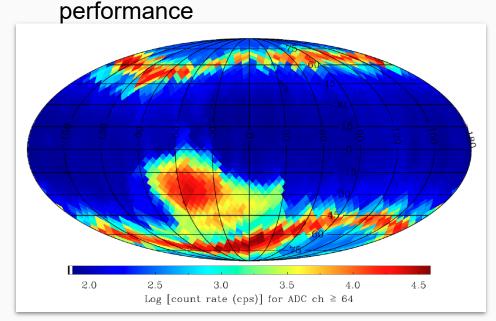
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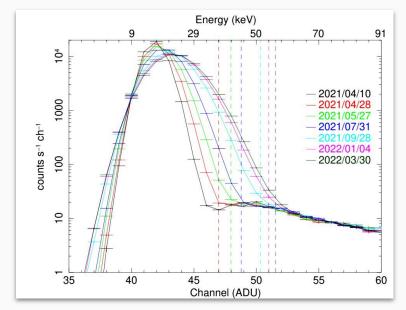
Radiation Environment



GRBalpha also provides useful information about radiation environment and MPPC



Count rate map by GRBalpha (particle background, CXB, albedo X-ray) => rate trigger algorithm



Noise spectrum as measured by GRB in one year operation (degradation of MPPC in LEO)

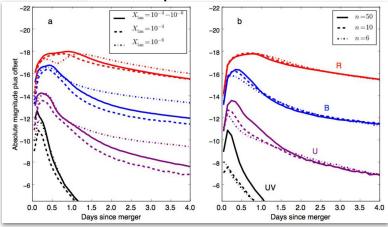


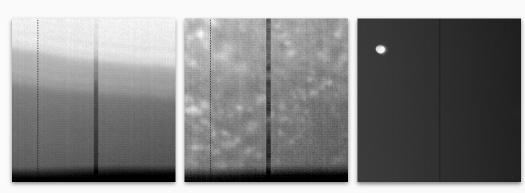
GRBbeta (2024.07-)



2U Cubesat proof-of-concept mission for a technology demonstration

- smaller effective area design to prove CAMELOT mission concept (all-sky monitoring and timing-based localization of GRB)
- experimental UV camera called LUVCam (precursor for future UV observations of kilonova),
- various platform-side features including attitude determination and control subsystem





UV image of earth rim, clouds and moon



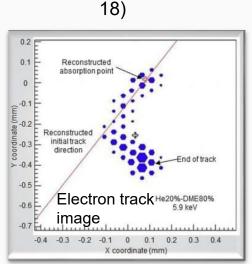
IXPE Instruments

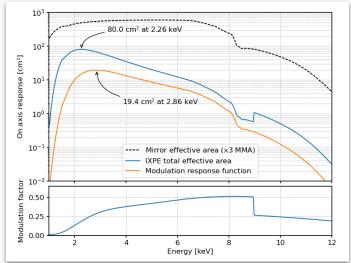


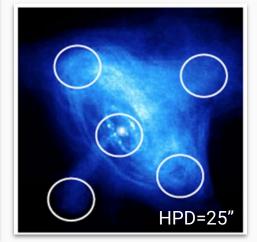
(see Soffitta+21 and Baldini+21 for latest information)

2-8 keV, 3 Mirror Module Assemblies (MMAs) and Detector Unites (DUs)

- MMAs: each contains 24 nested shells and has >200 cm² (3-6 keV)
- DUs: Gas pixel detector, measure photoelectron track (polarization) direction
 - FOV=12.9' x 12.9', HPD=25", m₁₀₀>0.5 achieved
 - o Event-by-event Stokes parameter to use imaging-polarimetry capability (Kislat+15, Vink & Zhou









Mission Status



Baseline mission completed successfully

- Almost all classes of sources observed; >70 discovery papers (3 in Nature, 2 in Science)
- Data are released 1 week after completion of obs.

GO phase started in 2024/Feb, cycle2 will be 2025/Feb-Aug

Call for proposals (incl. ToOs) just closed and being selected

• Unanticipated ToOs can be requested via the IXPE ToO website

	Category	Average Time per Source [ks]	Sources [#]	Observations [#]
	PWN	940	4	7
	SNR	800	5	7
1	Stellar BH	670	7	15
	NS LMXB	150	9	11
	Accreting Pulsar	420	9	17
	Magnetar	970	4	4
1	Blazar Radio Gal	390	12	17
1	Radio Quiet AGN Sgr A	820	5	6
1	GRB	100	1	1
1	Total	540	56	85

IXPE ToO observation requests will not be considered for events or sources that could have been predicted or proposed form until IXPE can slew to the target and start observing.	or in advance. If the ToO is accepted, it will take 3 calendar days or so from the time you submit thi
DXPE should not be used just to measure the X-ray flux of a source. DXPE is intended to measure the polarization of level of polarization you expect to see from your source. In any case, you must estimate the Minimum Detectable Polarizestimated using WebPIMMS.	
The ability to get data off the spacecraft is limited and this limits how long a bright source can be observed before we ne board storage is filled (assuming it was empty at the start) and it will take up to a week to download the data. Therefore	
Please review the IXPE Long Term Plan to see if your proposed target is not already listed.	
Please check to see if your target is currently observable with IXPE using viewing.	
In the first two years, we encourage the community to collaborate with the IXPE science team. If the mission is extende Principal request	a full GO program will be implemented.
IDPE data associated with ToO requests will have no exclusive use period and will be available via the public archive. In the first two years, we encourage the community to collaborate with the IMPE science team. If the mission is extende Principal request Name Institute	a full GO program will be implemented.
In the first two years, we encourage the community to collaborate with the IXPE science team. If the mission is extende Principal request Name Institute Primary Email address (additional email addresses can be supplied in Remarks section below). Note, if you do not get an email sent to this	a full GO program will be implemented.
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T. Mizuno 2025.11.19 19/1



Stokes Parameter Based Analysis



Event-by-event Stokes parameters:

• $i_k=1$, $q_k=2\cos 2\theta_k$, $u_k=2\sin 2\theta_k$

Stokes parameters of the entire data:

• $I=\Sigma i_k$, $Q=\Sigma q_k$, $U=\Sigma u_k$

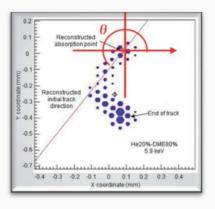
Normalized Stokes parameters, PD & PA:

• $Q_N=Q/I$, $U_N=U/I$, $PD=(1/m_{100})$ sqrt $(Q_N^2+U_N^2)$, PA=(1/2) arctan2(U, Q)

Erros:

• $V(Q)=\Sigma q_k^2$, $V(U)=\Sigma u_k^2$

Aeff, m_{100} , and reconstruction quality of each event can also be taken into account (unlike PD/PA, Stokes params. are additive and allow flexible binning in space and time)





For "Do It Yourself" Persons



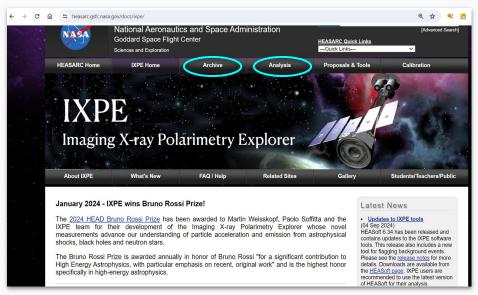
IXPE Data archived by NASA's HEASARC

Data format and HEASOFT analysis tool well documented

Alternative package (ixpeobssim) also available (link under GOF "Contributed IXPE

Software" page

Much of analysis can be done in Xspec





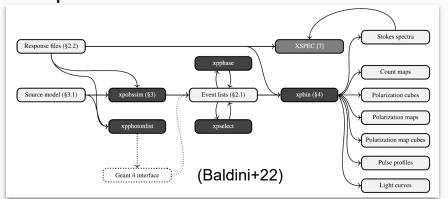
For "Do It Yourself" Persons (Cont'd)



You may use xselect to read/filter events and extract spectrum

```
xsel> read event "./ixpe01004701_det1_evt2_v01.fits.gz"
xsel> filter region "src.reg"
xsel> extract SPEC stokes=NEFF
xsel> save spec ixpe_det1_src_
```

Or, use ixpeobssim to read/select events and bin spectrum



xpselect --regfile src.reg --sufix sel ixpe01004701_det1_evt2_v01.fits

xpbin --algorithm PHA1Q --irfname ixpe:obssim:alpha075_v012 --weights True ixpe01004701_det1_evt2_v01_sel.fits

You will have 3 outputs: Stokes-I/Q/U spectra



For "Do It Yourself" Persons (Cont'd)

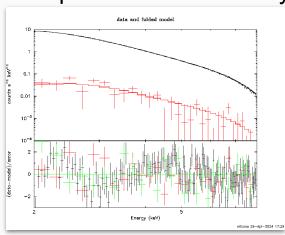


3 responses (not 2) required for each detector: rmf, arf, and mrf

- mrf = arf*µ₁₀₀
- use ixpecalcarf to generate arf/mrf

```
> ixpecalcarf \
evtfile=ixpe01004701_det1_evt2_v01.fits.gz \
attfile=ixpe01004701_det1_att_v01.fits.gz \
arfout=ixpe_det1_src_Q.mrf \
specfile=none radius=1.0 weight=1 resptype=mrf
```

mrf shall be read instead of arf for Stokes-Q or U spectra. Then you may fit 3 spectra simultaneously with, e.g., TBabs*polconst*powerlaw



Stokes-I (black)

Stokes-Q (red)

Stokes-U (green; negative and not shown in upper panel)

(Ixpeobssim may be more user-friendly for imaging-polarimetry analysis [like Vela PWN])



Note On Statistics



Since PD shall be >=0, PD-PA contour will be skewed when the significance is not so high (σ <=3)

If so, examine Stokes-Q/U plane instead of PD/PA (w/ ixpeobssim); error contours are circular and you can adequately evaluate significance and errors

• PD=sqrt($Q_N^2+U_N^2$), PA=(1/2) arctan2(U, Q)

Use ixpeobssim and Stokes-Q/U for imaging-polarimetry analysis

 See Kislat+15 and Vink&Zhou18 for the formalism (Mizuno+23 may also be useful)

